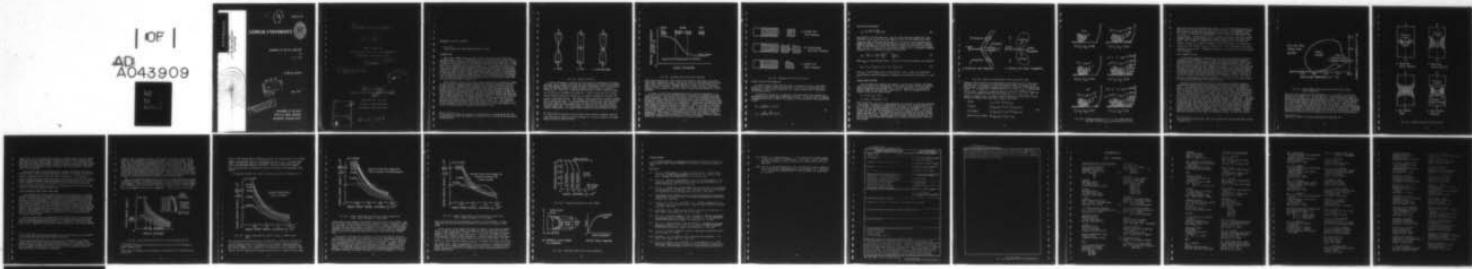


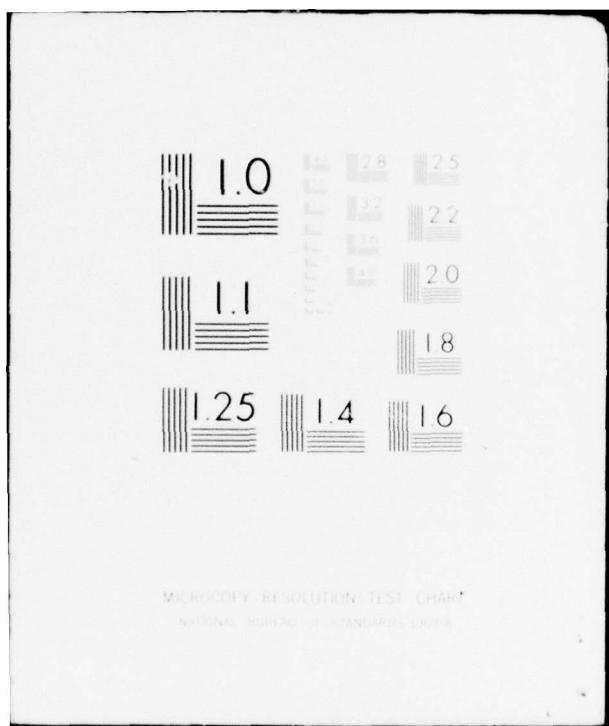
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MECHANICS OF DUCTILE FRACTURE

BY

G. C. SIH

TECHNICAL REPORT



JUNE 1977



DEPARTMENT OF THE NAVY
OFFICE OF NAVAL RESEARCH
ARLINGTON, VIRGINIA 22217

⑨ MECHANICS OF DUCTILE FRACTURE

by

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Bethlehem, Pennsylvania 18015

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MECHANICS OF DUCTILE FRACTURE

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INTRODUCTION

There are a number of ways in which a metal can deform and/or break into separate pieces. Even in a simple tensile test, wide variations can be found in the mode of failure depending on the material and operational conditions. Single crystals of soft metals such as lead and gold tend to neck and thin out to a fine point before separation takes place, Figure 1(a). This process of surface creation is more appropriately described as deformation rather than fracture and is relatively simple to analyze which will not be discussed further. At the other extreme for materials such as glass and hardened steels, very little deformation is observed before breaking, Figure 1(b). The more common types of fracture are those occurring in polycrystalline materials which show some necking or deformation and then break leaving fracture surfaces* with a characteristic shape such as a cup-and-cone shown in Figure 1(c).

The situation in Figure 1(c) is most complex and cannot be adequately treated by the classical fracture mechanics theory [1] originally proposed to explain brittle fracture. Deformation of moderately ductile metals is normally accomplished by the formation of tiny cavities or voids that eventually coalesce to form a macrocrack in a plane normal to the applied load, although this may be a zig-zag shape on a microscopic scale. As this crack moves towards the free surface, triaxial stresses are set up in the material and the specimen undergoes both shape change (distortion) and volume change (dilatation). This process continues until an instability point is reached at which time the crack runs out of the specimen at an angle to the original crack plane, Figure 1(c). This change of direction may either form the cone on the upper half and cup on the lower half of the specimen or vice versa. The choice depends on a small nonalignment of the crack with the mid-plane of loading as discussed in [2,3].

*The problem of whether the fracture is transgranular or intergranular will not be addressed here since the analysis to follow will not include the microstructure of the material.

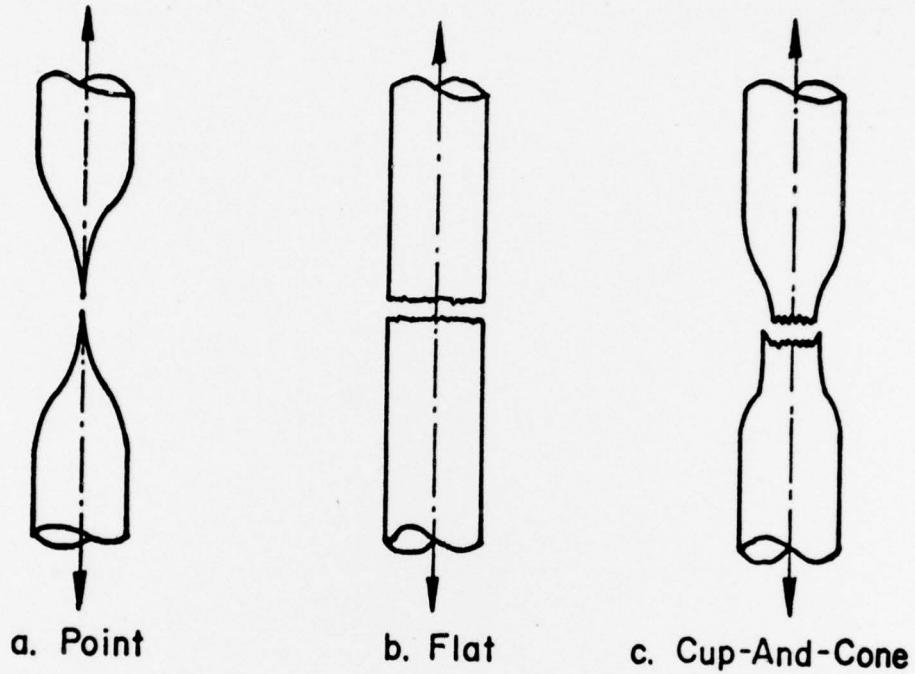


Fig. (1) - Types of fracture

A similar phenomenon is observed in the fracture of moderately thick plate specimens with a through crack such that the combinations of material properties and plate geometry would not satisfy the ASTM requirements [4] for plane strain where the smallest geometric length parameter must be greater or equal to $2.5(K_{Ic}/\sigma_{ys})^2$. Here, K_{Ic} is the valid critical Mode I stress intensity factor and σ_{ys} the yield strength of the material. Figure 2 shows that the critical stress intensity factor K_c^* becomes K_{Ic} or geometry independent only when the plate is sufficiently thick. The fracture surface appearance as shown in Figure 3(a) is almost all flat, i.e., the through crack grows in its initial plane normal to the applied load. As the plate thickness is decreased, the crack will initially tunnel and then deviate from its own plane near the plate surfaces forming "shear lips". The crack growth is stable at first advancing from its initial configuration in a planar fashion marked by the dotted area in Figure

* K_c should not be referred to as the plane stress fracture toughness value because the mode of fracture for this specimen is in no way related to the plane stress plasticity crack solution.

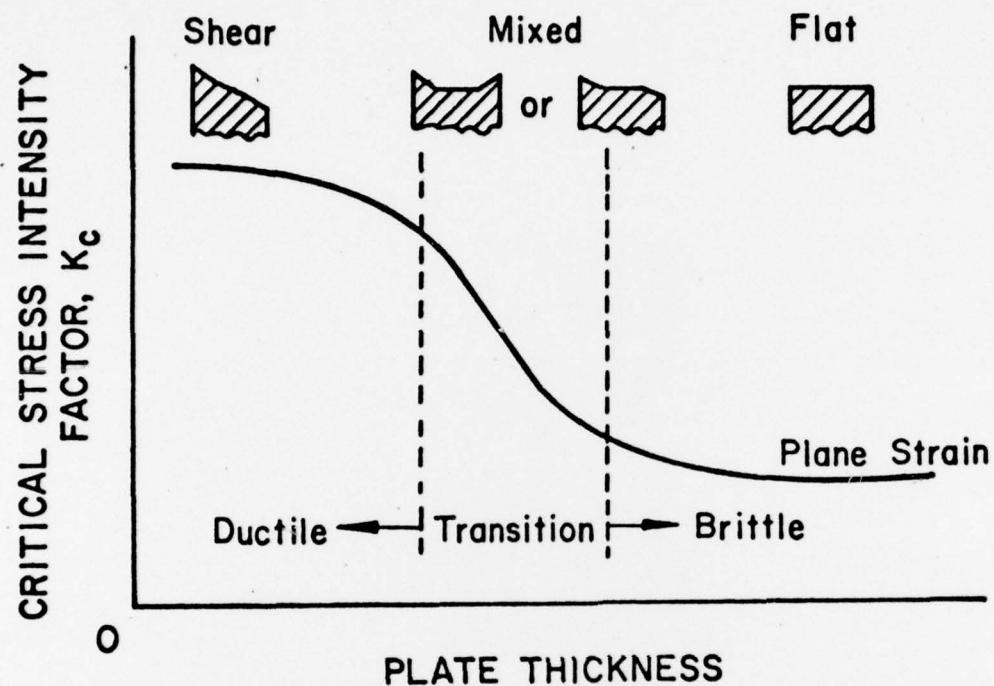


Fig. (2) - Variation of K_c with plate thickness

3(b) with a curved front. Upon reaching the point of instability, the curved crack breaks through the plate. This can be traced on the loading history curve. Again local conditions will determine whether the shear lip will form on one crack surface or the other. Further reduction in plate thickness leads to the decrease of flat surface, Figure 1(c), and a slanted fracture pattern is finally observed.

In order to predict the aforementioned ductile fracture phenomenon, it is necessary to have a fracture criterion that is amenable to analyzing mixed mode crack propagation. The regions of distortion must be distinguished from those undergoing large dilatation such that prediction on crack trajectory can be made. On physical grounds, distortion is associated with yielding of the material while dilatation is associated with crack propagation. Figures 4(a) and 4(b) show that under symmetric loading distortion or yielding is off to the sides of the crack and dilatation leading to fracture is directly ahead. A description of the ductile fracture process involves the interplay between yielding and crack growth which is fundamental to the understanding of transition from slow to rapid crack propagation. That problem is difficult because it not only requires a three-dimensional elastic-plastic analysis of a finite width and/or thickness specimen containing a growing crack but a realistic fracture criterion.

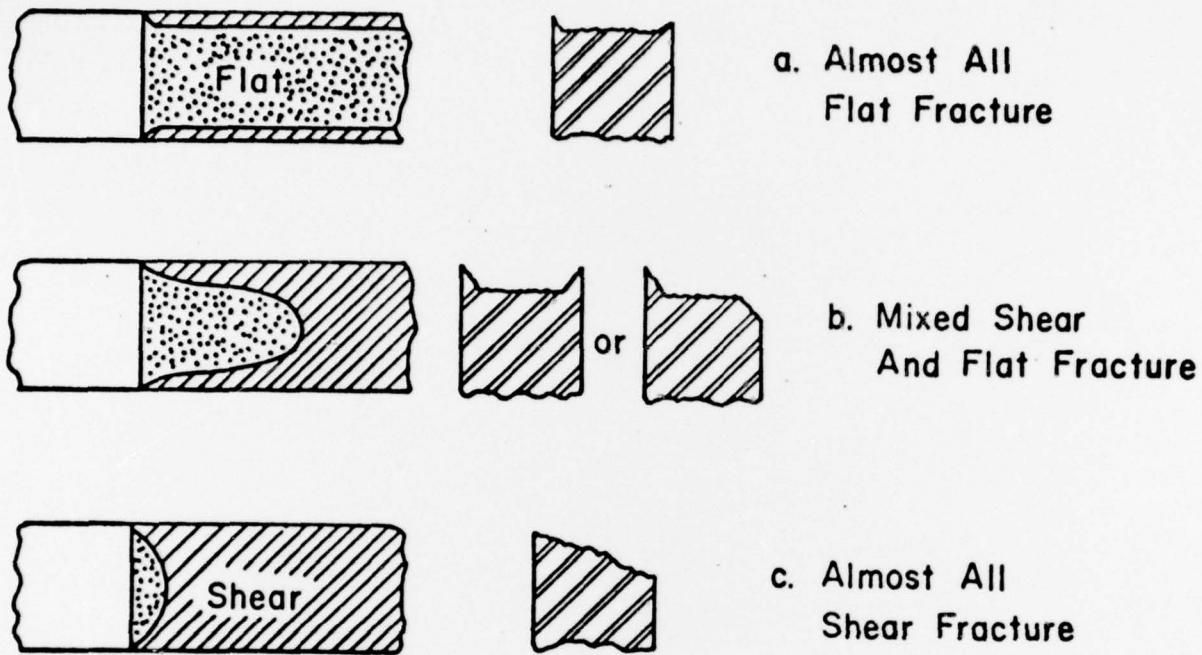


Fig. (3) - Appearance of fracture surfaces

ELASTIC-PLASTIC STRESS ANALYSIS

In order to gain insight into the shear lip formation process, a two-dimensional elastic-plastic stress analysis [5] is carried out to investigate the phenomenon of crack turning during the last ligament growth when it approaches the specimen surface.

The numerical calculation is based on a two-dimensional finite element program employing the twelve node isoparametric element. A special crack tip element is employed where use is made of the singularity solution [6] for the stresses and strains

$$\sigma_{ij} = \frac{K}{r^{1/(1+n)}} \sigma_{ys} f_{ij}(\theta) \quad (1)$$

$$\epsilon_{ij} = \frac{K^n}{r^{1/(1+n)}} \frac{\alpha \sigma_{ys}}{E} g_{ij}(\theta)$$

while the displacements

$$u_i = K^n r^{\nu(1+n)} \frac{\alpha \sigma_{ys}}{E} h_i(\theta) \quad (2)$$

are finite as $r \rightarrow 0$, the crack tip. When the power hardening exponent $n=1$, the functions $f_{ij}(\theta)$, $g_{ij}(\theta)$ and $h_i(\theta)$ reduce to those for the elastic case with θ being the polar angle measured from the line of expected crack extension under symmetric loading. For the elastic-plastic problem, the coefficient* K does not have the same meaning as stress (or strain) intensity factor used in the elastic case. In Eqs. (1) and (2), E is the Young's modulus and α is a material parameter employed in the J_2 deformation theory of plasticity:

$$\epsilon_{ij} = \frac{1+\nu}{E} s_{ij} + \frac{3}{2} \frac{\alpha}{E} \left[\left(\frac{\sigma_{eff}}{\sigma_{ys}} \right)^{n-1} - \frac{\sigma_{ys}}{\sigma_{eff}} \right] s_{ij} \quad (3)$$

where σ_{eff} is the effective stress. The strain and stress deviators are given by

$$\epsilon_{ij} = \epsilon_{ij} - \frac{1}{3} \epsilon_{pp} \delta_{ij}, \quad s_{ij} = \sigma_{ij} - \frac{1}{3} \sigma_{pp} \delta_{ij} \quad (4)$$

with $\epsilon_{pp} = [(1-2\nu)/E]\sigma_{pp}$ and ν is the Poisson's ratio. Hence, the particular elastic-plastic material behavior is completely determined by specifying E , α and n .

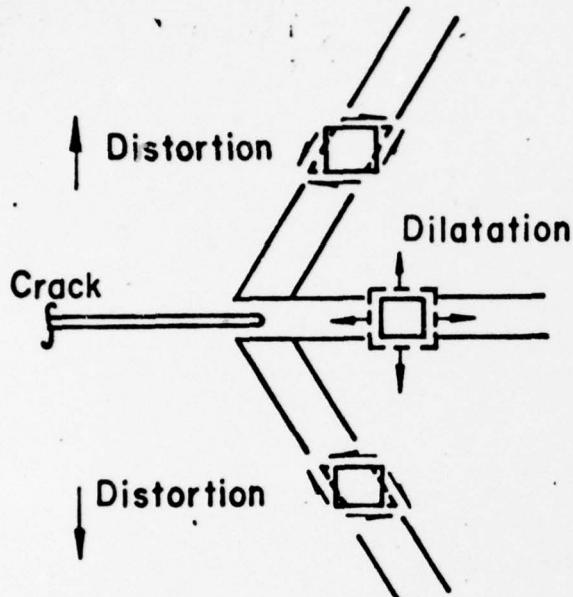
CENTER CRACK SPECIMEN

For the purpose of numerical calculation, a center crack specimen with dimensions of 4 in by 4 in is considered. The specimen contains a crack of length $2a$ while $2b$ designates the specimen width. The far field stress is oriented uniaxially normal to the crack with a magnitude σ_0 . Material parameters are selected to take the following values:

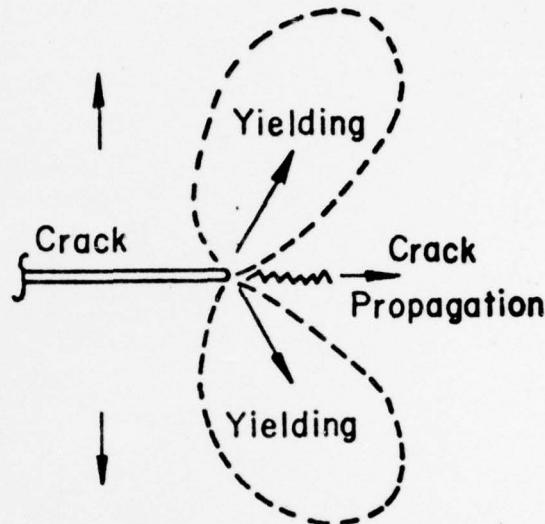
$$E = 1 \text{ psi}, \quad \sigma_{ys} = 1 \text{ psi}$$

$$\alpha = 0.5, \quad \nu = 0.3 \text{ and } n = 5.0$$

*The choice of using K , K^n or some combination of K and K^n as a fracture criterion is not clear. It should be cautioned that the form of the singular solution in Eqs. (1) and (2) was made possible only by assuming that the material in a small circular or core region surrounding the crack tip yields uniformly. This is obviously not the case. The consequence of such an assumption as to how it would affect the results in a qualitative and quantitative manner has not been studied and is still not known. Hopefully, the finite element solution based on Eqs. (1) and (2) would not be affected seriously at a distance sufficiently far from the core region.



a. Distortion And Dilatation



b. Yielding And Crack Propagation

Fig. (4) - Zones of distortion and dilatation ahead of crack

Figures 5 show a central crack penetrating through eighty percent of the specimen width ($a/b = 0.8$) while its side has necked in about five percent of the crack length. The different degrees and pattern of yield are illustrated by the sequence of line drawings in Figures 5(a) to 5(f) in which σ_o/σ_{ys} is varied from 0.20 to 0.42. Yielding is assumed to take place when the effective stress σ_{eff} in a material element exceeds the yield stress σ_{ys} , i.e., $\sigma_{eff} > \sigma_{ys}$. The squares, triangles, etc., are used to signify the various intensities of yielding and the ranges within which they cover are defined as

$$\begin{aligned}
 \text{Square:} \quad & \sigma_{ys} \leq \sigma_{eff} < \frac{1}{4} (\sigma_{max} - \sigma_{ys}) \\
 \text{Triangle:} \quad & \frac{1}{4} (\sigma_{max} - \sigma_{ys}) \leq \sigma_{eff} < \frac{1}{2} (\sigma_{max} - \sigma_{ys}) \\
 \text{Plus symbol:} \quad & \frac{1}{2} (\sigma_{max} - \sigma_{ys}) \leq \sigma_{eff} < \frac{3}{2} (\sigma_{max} - \sigma_{ys}) \\
 \text{Multiplication symbol:} \quad & \frac{3}{2} (\sigma_{max} - \sigma_{ys}) \leq \sigma_{eff} \leq \sigma_{max}
 \end{aligned} \tag{5}$$

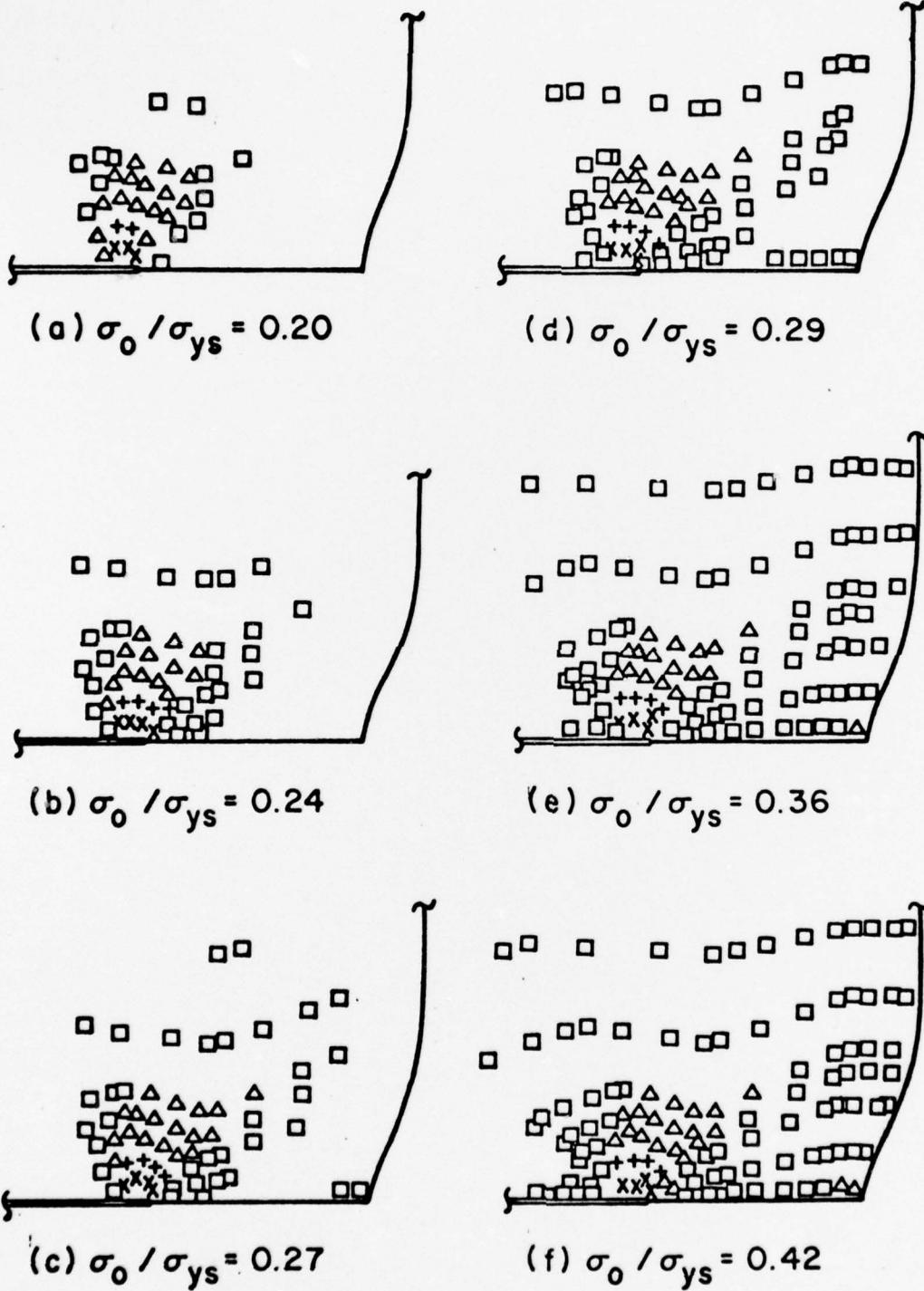


Fig. (5) - Yielding pattern of a 4 in x 4 in cracked specimen with $a/b = 0.8$ and σ_0 / σ_{ys} varied from 0.20 to 0.42

Note that the plus and multiplication symbols represent intensified yielding and they occur in a region confined close to the crack tip. The intensity of yielding decreases with distances away from the crack tip until no yielding occurs. For $\sigma_0/\sigma_{ys} = 0.20$ and 0.24 , the specimen is still elastic at large except for a localized zone. As the applied load is increased, the region next to the necked boundary begins to deform plastically leaving a path of elastic material at an angle. This is shown in Figures 5(c) and 5(d) where $\sigma_0/\sigma_{ys} = 0.27$ and 0.29 . Further increase in σ_0 leads to yielding of the entire section as indicated in Figures 5(e) and 5(f) for $\sigma_0/\sigma_{ys} = 0.36$ and 0.42 . It is important to note that yielding is not uniform. The higher intensities occur near the crack tip and the necked boundary.

LAST LIGAMENT FAILURE

The pertinent question is whether the crack would run through the region of maximum plastic deformation (i.e., highest intensity of yielding) toward the specimen boundary or would attempt to avoid the deformed material, growing into the elastic zone in front of the tip and then diverging from the line of symmetry to break through the last ligament at an angle*. The strain energy density criterion [7] has been applied to predict the phenomenon of mixed mode fracture. The path of minimum strain energy density consistently lies between the region of maximum plastic yielding and the horizontal axis that connects the crack tip to the free surface for several specimen configurations. Figure 6 gives the result for a typical calculation where $\sigma_0/\sigma_{ys} = 0.28$ and $a/b = 0.60$. The theory predicts crack propagation along a path next to the inner elastic-plastic boundary intersecting the surface at an oblique angle.

Although the foregoing results must be regarded as preliminary, they are nevertheless helpful in gaining insights into the ductile fracture behavior of metals. An interpretation of fracture modes associated with the relative degrees of material ductility is given in Figures 7(a) to 7(d). The fracture path for a relatively brittle material is mostly flat with yielding localized to the crack tip region, Figure 7(a). In such a case, energy is dissipated predominantly due to the creation of a free or crack surface. The more ductile behavior will exhibit appreciable deformation through necking as in Figure 7(b). The path of fracture will tend to curve as the specimen boundary is approached. Since both Figures 7(a) and 7(b) show that crack extension prevails in the elastic portion of the material, the energy required to extend a unit crack extension at global instability must then be the same. This means that the same fracture toughness value must apply to the same material that exhibits either brittle or ductile behavior owing to changes in specimen size. In other words, material constants should not be sensitive to changes in geometry and size of specimens. A more detailed discussion on this topic can be found in [8,9]. For the very ductile

* In a cylindrical tensile bar, this last ligament forms the cup-and-cone as shown in Figure 1(c).

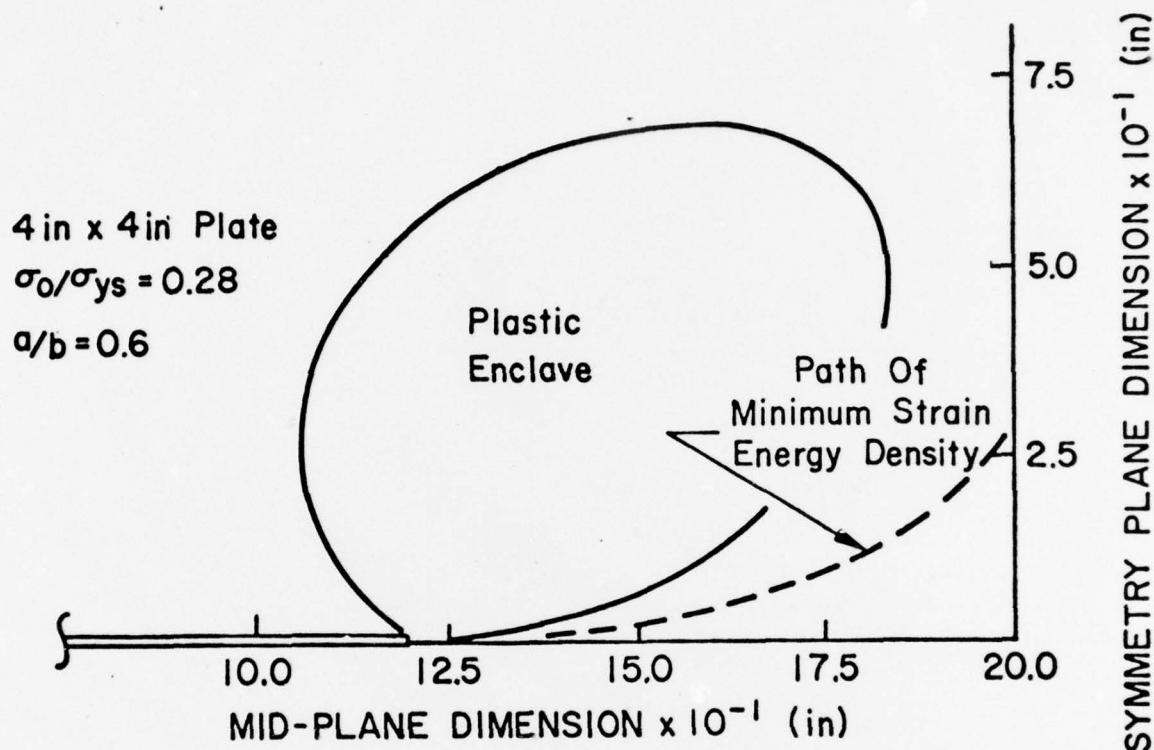
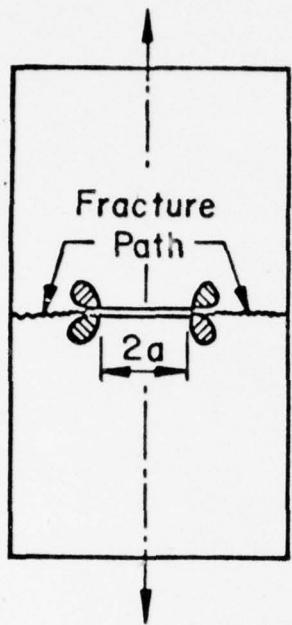


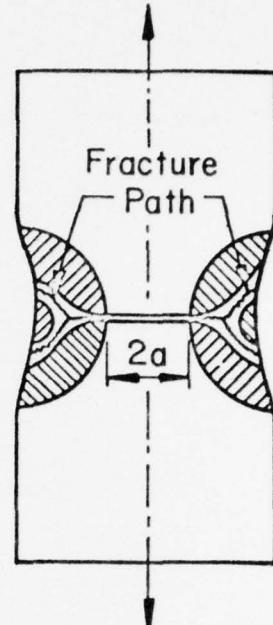
Fig. (6) - Last ligament failure predicted by the strain energy density criterion

materials* that undergo extensive deformation, it is possible to achieve net section yield, Figure 7(c). In that case, the fracture pattern will again be developed in accordance with the distribution of distortion and dilatation which can be predicted from the strain energy density theory. This is being referred to as the second degree of yielding in Figure 7(d). Refer to Figure 5(f) for the details of an actual elastic-plastic stress calculation. The specimen will of course be necked considerably by this time and the cup-and-cone when developed will be much smaller in size than that of the material in Figure 7(b). Continuum mechanics analysis is limited to sorting out the regions of distortion (yielding) from those of dilatation (crack growth as assumed in the strain energy density

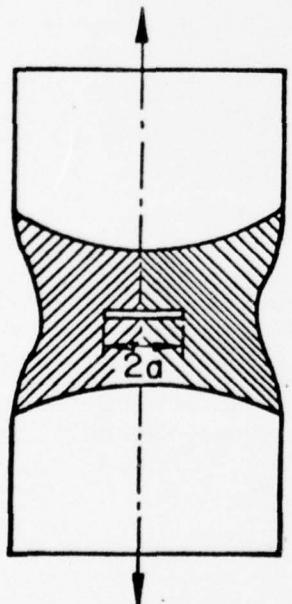
*Ductility is known to increase with decreasing specimen size.



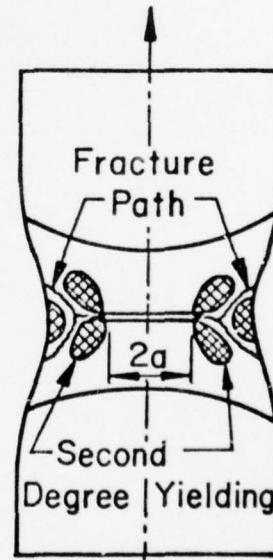
a. Relatively
Brittle Behavior



b. Relatively
Ductile Behavior



c. Net Section
Yield



d. Very Ductile
Behavior

Fig. (7) - Brittle and ductile fracture modes

theory [7]). It is the proportion of energy dissipated through plastic deformation (micro surface creation) and fracture (macro free-surface creation) that determines the final fracture mode. Generally speaking, it is desirable to balance these two energy dissipations and hence structural members are often called to operate in the transition region, Figure (2) for plate specimens and Figure 7(b) for bar specimens.

An extensive amount of experimental work has been done [10] in the past to show that the critical strain energy density $(dW/dV)_c$ is characteristic of the material undergoing both elastic and plastic deformation. Once $(dW/dV)_c$ is known, analysis may be carried out to predict the allowable load and net section size of structural members that deform elastically and/or plastically. The relation $(dW/dV)_c = S_c/r_0$ can also be used to locate the distance r_0 from the crack front at which failure initiates. The quantity S_c has been known as the critical strain energy density factor [7].

THREE DIMENSIONAL CRACK GROWTH PREDICTIONS

A three-dimensional finite element program for a through crack in a finite thickness plate has been developed [11] to examine the phenomenon of crack tunneling. The crack front may either be straight or curved depending on the constraint* of the plate surfaces. Calculations have been made for various crack front shapes in conjunction with crack growth modeling based on the strain energy density criterion. The stress distribution near the crack front is found to be sensitive to the details of the specimen geometry, material properties and loading. This suggests that the crack may adapt itself into a natural shape at which unstable fracture can occur**. Information on the size and shape of this crack at instability is pertinent to resolving the little understood problem of thickness size effect.

The aim of this work is to model the stable growth process in a ductile material by separately analyzing the influences of crack front geometry and the contributions from material nonlinearity. The influence of crack front curvature is first examined with the aid of a three-dimensional elastic finite element

* In the more ductile material where shear lips are developed the crack profile can be curved and grow appreciably before instability.

** Experimental observations [13] in the fracture of moderately thick plates loaded monotonically show that stable crack growth is first initiated in the specimen interior thereby increasing the front curvature. Instability or rapid crack propagation is observed to occur at about the time when the shear lips (nonplanar crack front growth) are developed.

analysis* [12] in conjunction with a local growth criterion to predict increments of crack growth and corresponding changes in crack front shape. In general, both the direction and magnitude of local crack growth from the current crack front are expected to vary with position along that front. The strain energy density field surrounding the crack front is the basis for the development of the local crack growth criterion [8,14]. It is postulated that the path of growth from each point along the crack edge will follow the minimum strain energy density path emanating from that point. Further, growth at points along the current crack front will initiate when the strain energy density at a "core" distance r_0 along the minimum path reaches a preset value $(dW/dV)_c$.

Hence, the continuous growth of the crack front is approximated by discrete increments of growth. The amount of growth at a point along the crack front in an increment is taken to be the distance along the minimum strain energy density path, $(dW/dV)_{\min}$, to the point where $(dW/dV)_{\min}$ reaches a preset value**, $(dW/dV)_c$, which depends on the material properties. Figure 8 gives a descriptive

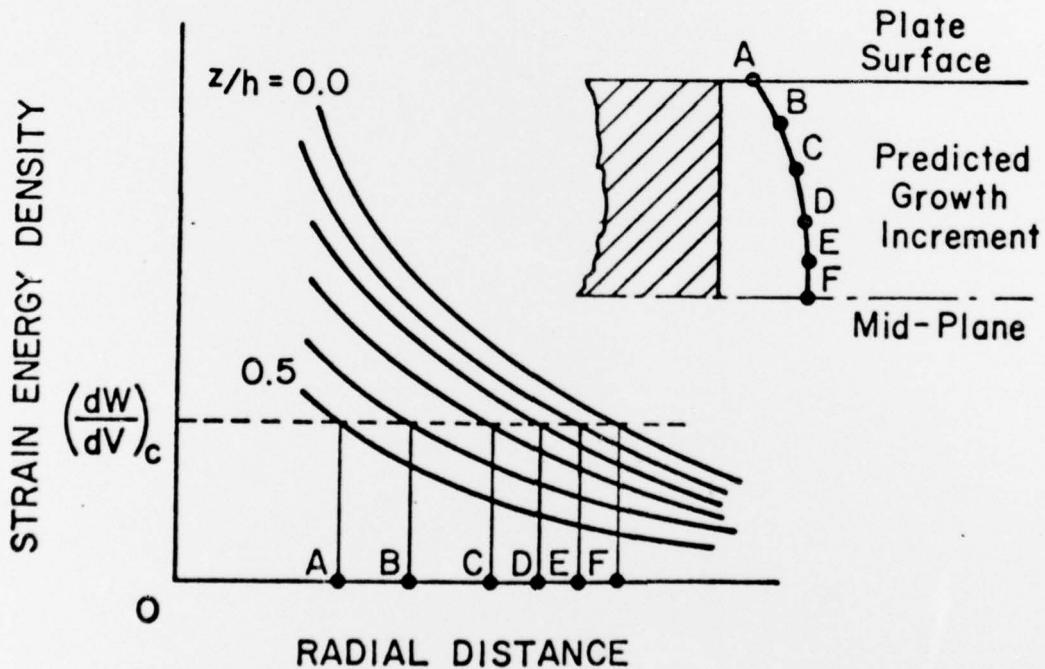


Fig. (8) - Crack growth prediction based on strain energy density

* The effect of material nonlinearity due to plasticity will be presented in another communication.

** This value of $(dW/dV)_c$ can be determined experimentally for a particular material [11].

plot of strain energy density $(dW/dV)_{\min}$ against the radial distance r in a plane normal to the applied load from the associated points A, B,---, F on the crack front. The new crack profile is determined from the various values of r and z/h on the graph where $(dW/dV)_c$ intersects. The selected value of $(dW/dV)_c$ will obviously affect the magnitude of growth and shape of the subsequent crack fronts.

The numerical results for a specific example are shown in Figures 9 to 11.

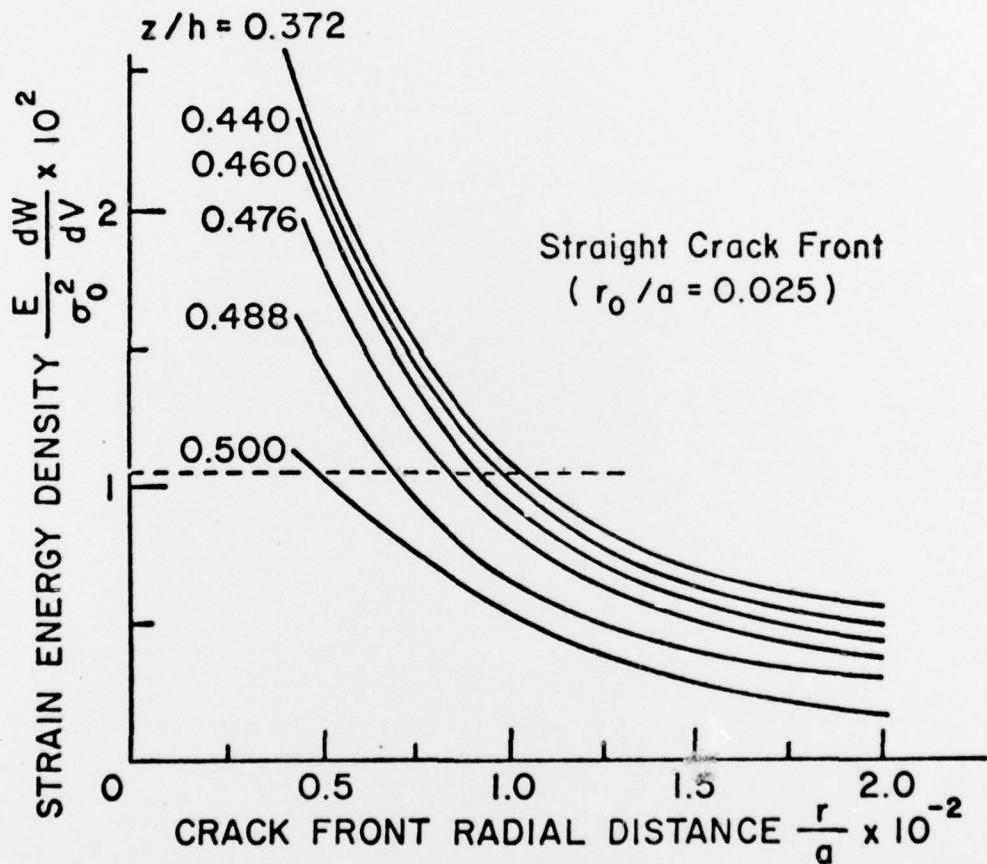


Fig. (9) - Strain energy density variation along a straight crack front

The normalized strain energy density $(E/\sigma_0^2)(dW/dV)$ is plotted as a function of r/a for different values of z/h where z is the thickness coordinate measured from the mid-plane. $z/h = 0.5$ locates the plate surface. Note that for a pre-set value of $(dW/dV)_c$ less changes in r are observed as z/h is decreased. This implies that the crack profile undergoes relatively small change in the specimen

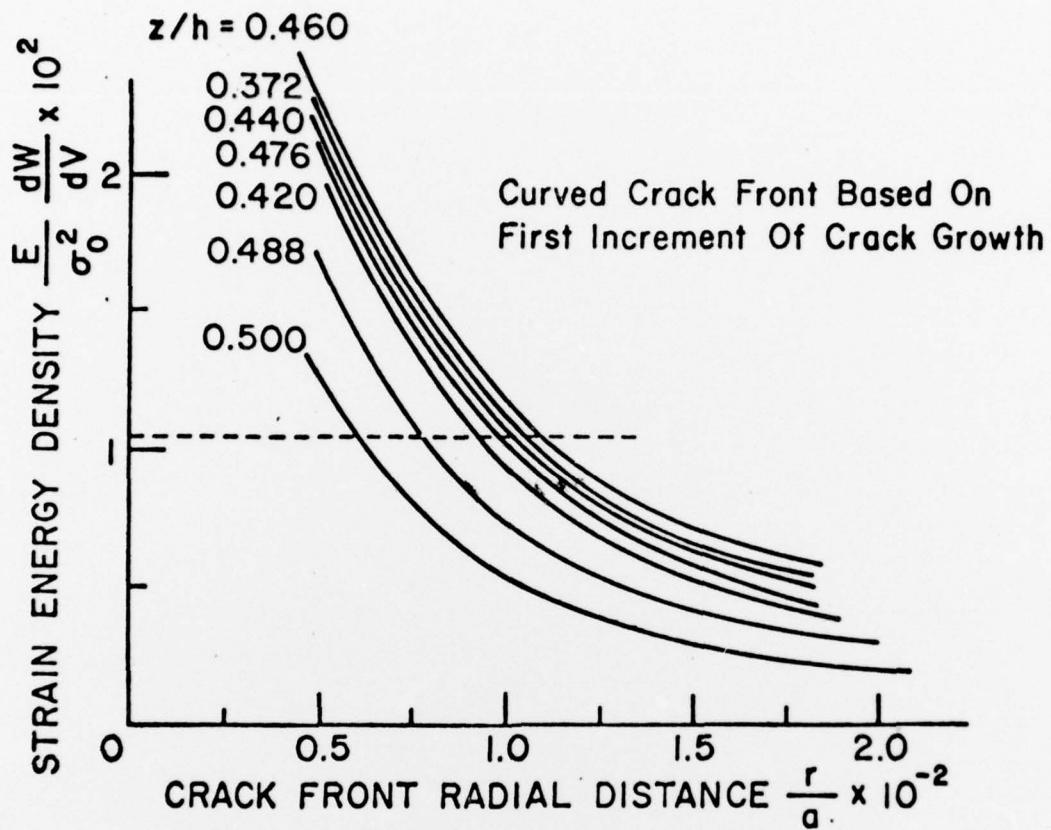


Fig. (10) - Strain energy density variation along a curved crack front - first increment of crack growth

interior while large changes are observed in regions close to the surface. Figure 12 displays the predicted crack front geometry with each increment of growth. The load level is kept constant during this period. The more pronounced crack curvature near the plate surface is indicative of the fact that the distortional component of strain energy density increases as the free surface is approached. This increase occurs even more rapidly for curved crack fronts than straight ones.

The results in Figures 9 to 12 explain in part why the material near the plate surface is more likely to yield and to cause the development of shear lips which in turn provide the constraint to contain crack growth. Crack tunneling is a consequence of this surface constraint, Figure 13(a). The point at which the crack overcomes this constraint corresponds to global instability, Figure 13(b), of the specimen; that is, when the specimen loses its structural integrity. It should be cautioned that nonlinear load-displacement response in the global sense may not always correspond to material nonlinearity. In this

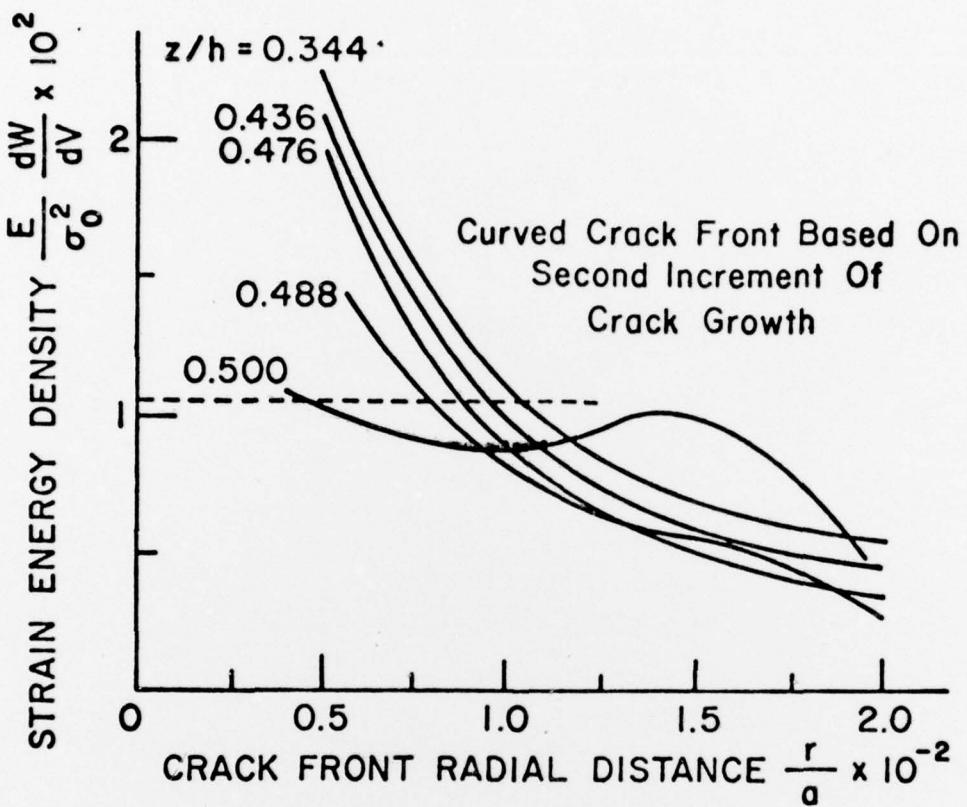


Fig. (11) - Strain energy density variation along a curved crack front - second increment of crack growth

case, the influence can be attributed to changes in crack geometry. As mentioned earlier, the contribution of material nonlinearity is currently under study.

At this point, a few remarks with regard to adding a crack-length increment (that is equal to the calculated one-dimensional plastic-zone size) onto the original crack length may be in order. This is the usual procedure in fracture mechanics to correct for plasticity effects. In view of the results of this investigation, it might be more appropriate to add an average half crack length a_c as in Figure 13(a) with $2(a+a_c)$ being the effective crack length. The correction a_c can be obtained simply by equating the shaded area inside and outside the tunnel crack. Once the phenomenon of ductile fracture is understood and the shape and size of these tunneling cracks are calculated for a variety of situations, there is no difficulty in establishing simple engineering formulas for design applications.

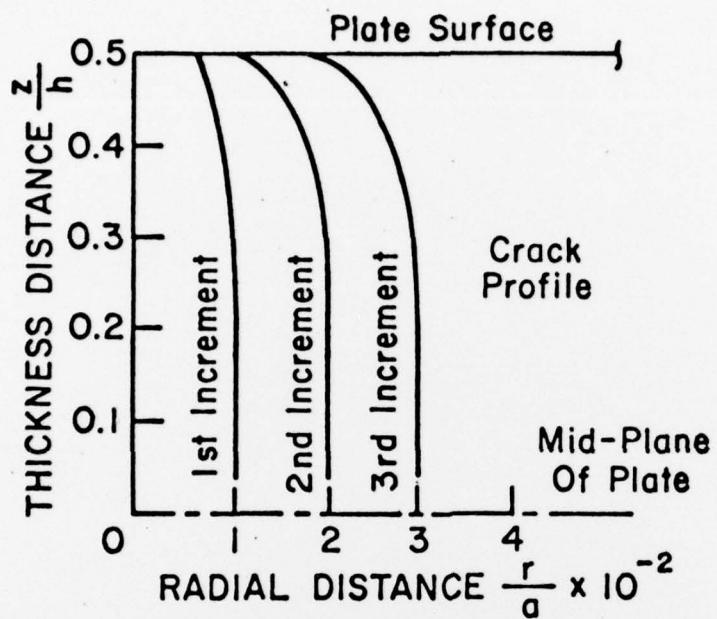
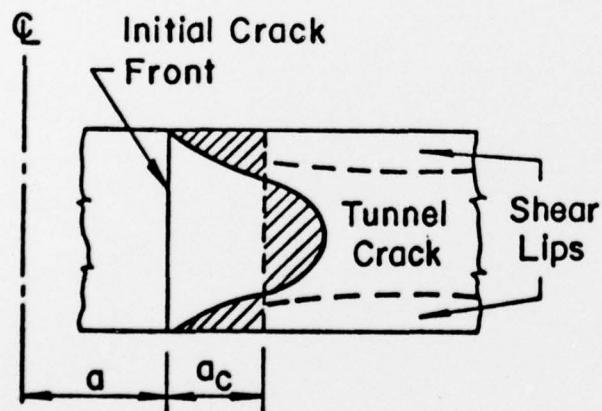
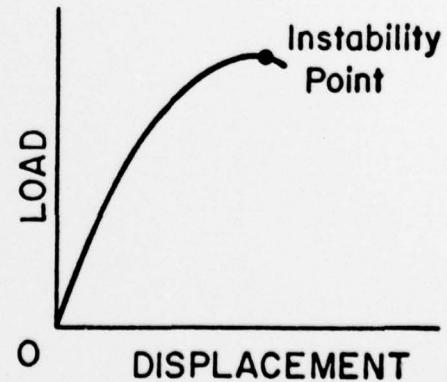


Fig. (12) - Predicted increments of crack growth



(a) Effective crack length at instability



(b) Non-linear response

Fig. (13) - Nonlinear effect due to crack tunneling

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REFERENCES

- [1] Sih, G. C. and Erdogan, F., "Stress intensity factor: concept and application", Linear Fracture Mechanics, edited by G. C. Sih, R. P. Wei and F. Erdogan, Envo Publishing Co., Inc., p. 85, 1975.
- [2] Sih, G. C., "Effect of nonalignment of load on crack propagation", Transactions of the 3rd International Conference on Structural Mechanics in Reactor Technology, Vol. 5, p. L718, 1975.
- [3] Sih, G. C., "Application of the strain energy density theory to fundamental fracture problems", Proceedings of the 10th Annual Meeting of the Society of Engineering Science, Raleigh, North Carolina, p. 221, 1973.
- [4] Plane Strain Crack Toughness Testing of High Strength Metallic Materials, edited by W. F. Brown, Jr. and J. E. Srawley, ASTM Special Technical Publication, No. 410, 1966.
- [5] Hilton, P. D., Sih, G. C. and Kiefer, B. V., "Growth characteristics of a through crack in a plate specimen", Technical Report IFSM-76-76, Lehigh University, June 1975.
- [6] Hutchinson, J. W., "Plastic stress and strain fields at a crack tip", J. Mech. Physics and Solids, Vol. 16, p. 337, 1968.
- [7] Sih, G. C., "A special theory of crack propagation", Methods of Analysis and Solutions of Crack Problems, Vol. 1, edited by G. C. Sih, Noordhoff International Publishing, Leyden, The Netherlands, p. 21, 1973.
- [8] Sih, G. C., "Elastic-plastic fracture mechanics", Prospects of Fracture Mechanics, edited by G. C. Sih, H. C. van Elst and D. Broek, Noordhoff International Publishing, Leyden, The Netherlands, p. 613, 1974.
- [9] Sih, G. C., "Fracture toughness concept", ASTM Special Technical Publication, No. 605, p. 3, 1976.
- [10] Gillemot, L. F., "Criterion of crack initiation and spreading", J. of Engin. Fract. Mech., Vol. 8, p. 239, 1976.
- [11] Sih, G. C., Hilton, P. D., Hartranft, R. J. and Kiefer, B. V., "Three-dimensional stress analysis of a finite slab containing a transverse central crack", Technical Report IFSM-75-69, Lehigh University, June 1975.

- [12] Miller, G. A. and Stephenson, E. T., "The effect of the carbide phases on the crack growth of 0.5% Mo-B steels under impact, cyclic, and monotonically increasing loading", Metallurgical Transactions, Vol. 5, p. 659, 1974.
- [13] Sih, G. C., "A three-dimensional strain energy density factor theory of crack propagation", Three-Dimensional Crack Problems, Vol. 2, edited by G. C. Sih, Noordhoff International Publishing, Leyden, The Netherlands, p. 15, 1975.

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The phenomenon of ductile fracture in plate specimens is described by a finite element stress analysis coupled with the strain energy density fracture criterion. The last ligament of failure tends to curve developing a cup-and-cone type of fracture in tensile bars and sheer lips in plate specimens. Three-dimensional crack growth profiles are developed by assuming that materials tend to break along the path of minimum strain energy density whose critical value is characteristic of the material and has been		

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measured experimentally for both elastic and plastic materials. The results are encouraging in that ductile fracture behavior can be modeled by holding the strain energy density constant. Hence, allowable load and net section size can be predicted for specimens that deform in the plastic range.

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